

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2020–14–03 The Boeing Company:**  
Amendment 39–21157; Docket No. FAA–2020–0097; Product Identifier 2019–NM–208–AD.

**(a) Effective Date**

This AD is effective August 27, 2020.

**(b) Affected ADs**

None.

**(c) Applicability**

(1) This AD applies to all The Boeing Company Model 737–300, –400, and –500 series airplanes, certificated in any category.

(2) Installation of Supplemental Type Certificate (STC) ST01219SE does not affect the ability to accomplish the actions required by this AD. Therefore, for airplanes on which STC ST01219SE is installed, a “change in product” alternative method of compliance (AMOC) approval request is not necessary to comply with the requirements of 14 CFR 39.17.

**(d) Subject**

Air Transport Association (ATA) of America Code 57, Wings.

**(e) Unsafe Condition**

This AD was prompted by a report that a crack indication consistent with fatigue cracking was found on the left nacelle support overwing fitting flange fastener hole during teardown of a Model 737–300 series airplane. The FAA is issuing this AD to address the potential for undetected cracks in the nacelle support overwing fittings or strut to wing diagonal brace, which could result in the inability of the structure to carry limit load and could adversely affect the structural integrity of the airplane.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

Except as specified by paragraph (h) of this AD: At the applicable times specified in the “Compliance” paragraph of Boeing Alert

Requirements Bulletin 737–57A1345 RB, dated December 17, 2019, do all applicable actions identified in, and in accordance with, the Accomplishment Instructions of Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019. Actions identified as terminating actions in Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019, terminate the applicable required actions of this AD, provided the terminating action is done in accordance with the Accomplishment Instructions of Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019.

*Note 1 to paragraph (g):* Guidance for accomplishing the actions required by this AD can be found in Boeing Alert Service Bulletin 737–57A1345, dated December 17, 2019, which is referred to in Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019.

**(h) Exceptions to Service Information Specifications**

(1) Where Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019, uses the phrase “the original issue date of Requirements Bulletin (RB) 737–57A1345 RB,” this AD requires using “the effective date of this AD.”

(2) Where Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019, specifies contacting Boeing for repair instructions, this AD requires doing the repair before further flight using a method approved in accordance with the procedures specified in paragraph (i) of this AD.

**(i) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Los Angeles ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (j)(1) of this AD. Information may be emailed to: 9-ANM-LAACO-AMOC-Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by The Boeing Company Organization Designation Authorization (ODA) that has been authorized by the Manager, Los Angeles ACO Branch, FAA, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

**(j) Related Information**

(1) For more information about this AD, contact Wayne Ha, Aerospace Engineer, Airframe Section, FAA, Los Angeles ACO Branch, 3960 Paramount Boulevard,

Lakewood, CA 90712–4137; phone: 562–627–5238; fax: 562–627–5210; email: wayne.ha@faa.gov.

(2) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (k)(3) and (4) of this AD.

**(k) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Requirements Bulletin 737–57A1345 RB, dated December 17, 2019.

(ii) [Reserved]

(3) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110–SK57, Seal Beach, CA 90740–5600; telephone 562–797–1717; internet <https://www.myboeingfleet.com>.

(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on July 6, 2020.

**Gaetano A. Sciortino,**

*Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2020–15818 Filed 7–22–20; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA–2020–0204; Product Identifier 2018–SW–082–AD; Amendment 39–21179; AD 2020–15–16]**

**RIN 2120–AA64**

**Airworthiness Directives; Leonardo S.p.A (Type Certificate Previously Held by Agusta S.p.A) Helicopters**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is superseding Airworthiness Directive (AD) 2018–07–08, which applied to certain Leonardo

S.p.A (type certificate previously held by Agusta S.p.A) Model A109E, A109K2, A109S, AW109SP, A119, and AW119 MKII helicopters. AD 2018–07–08 required reducing the life limit of the tail rotor blade retention bolt and an inspection of that bolt for cracking, and replacement of any cracked bolt. This AD continues to require reducing the life limit of the tail rotor blade retention bolt, inspecting that bolt for cracking, and replacing any cracked bolt. In addition, this AD requires repetitive inspections of the tail rotor blade retention bolt for cracking. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective August 27, 2020.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of April 26, 2018 (83 FR 15495, April 11, 2018).

**ADDRESSES:** For service information identified in this final rule, contact Leonardo S.p.A. Helicopters, Emanuele Bufano, Head of Airworthiness, Viale G. Agusta 520, 21017 C. Costa di Samarate (Va) Italy; telephone +39 0331 225074; fax +39 0331 229046; or at <https://www.leonardocompany.com/en/home>. You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110. It is also available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0204.

#### Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0204; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Matt Fuller, AD Program Manager, Continued Operational Safety Branch, Airworthiness Products Section, General Aviation and Rotorcraft Unit, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5151; email [matthew.fuller@faa.gov](mailto:matthew.fuller@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to supersede AD 2018–07–08, Amendment 39–19239 (83 FR 15495, April 11, 2018) (“AD 2018–07–08”). AD 2018–07–08 applied to certain Leonardo S.p.A Model A109E, A109K2, A109S, AW109SP, A119, and AW119 MKII helicopters. The NPRM published in the **Federal Register** on March 23, 2020 (85 FR 16281). The NPRM was prompted by the FAA’s determination that repetitive inspections of the tail rotor blade retention bolt are needed to address the unsafe condition. The NPRM proposed to continue to require reducing the life limit of the tail rotor blade retention bolt, inspecting that bolt for cracking, and replacing any cracked bolt. The NPRM also proposed to require repetitive inspections of the tail rotor blade retention bolt for cracking. Since issuing AD 2018–07–08, the FAA has determined that repetitive inspections of the tail rotor blade retention bolt are needed to address the unsafe condition. The FAA is issuing this AD to address cracked bolts, which could result in failure of the tail rotor and loss of control of the helicopter.

The European Aviation Safety Agency (now European Union Aviation Safety Agency) (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Emergency AD, 2016–0173–E, dated August 24, 2016 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Leonardo S.p.A. Model A109E, A109K2, A109LUH, A109S, A119, AW109SP and AW119 MKII helicopters. You may examine the MCAI in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2020–0204.

See the MCAI for additional background information.

##### Comments

The FAA gave the public the opportunity to participate in developing this final rule. The FAA received no comments on the NPRM or on the determination of the cost to the public.

##### Conclusion

The FAA reviewed the relevant data and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

#### Related Service Information Under 1 CFR Part 51

This AD requires the following service information, which the Director of the Federal Register approved for incorporation by reference as of April 26, 2018 (83 FR 15495, April 11, 2018).

- Leonardo Helicopters Mandatory Bollettino Tecnico No. 109EP–149, dated August 19, 2016.
- Leonardo Helicopters Mandatory Bollettino Tecnico No. 109K–72, dated August 19, 2016.
- Leonardo Helicopters Mandatory Bollettino Tecnico No. 109S–072, dated August 19, 2016.
- Leonardo Helicopters Mandatory Bollettino Tecnico No. 109SP–105, dated August 19, 2016.
- Leonardo Helicopters Mandatory Bollettino Tecnico No. 119–080, dated August 19, 2016.

This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### Differences Between This AD and the MCAI or Service Information

The MCAI does not specify life limits for a tail rotor blade retention bolt having part number (P/N) 709–0160–57–101 that has been interchanged between model helicopter installations, while this AD does.

The MCAI applies to Model A109LUH helicopters. Model A109LUH helicopters are not certified by the FAA and are not included on the U.S. type certificate data sheet; this AD therefore does not include those helicopters in the applicability.

#### Interim Action

The FAA considers this AD to be an interim action. The design approval holder is currently developing a modification that will address the unsafe condition identified in this AD. Once this modification is developed, approved, and available, the FAA might consider additional rulemaking.

#### Costs of Compliance

The FAA estimates that this AD affects 219 helicopters of U.S. registry. The FAA estimates the following costs to comply with this AD:

## ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Retained actions from AD 2018–07–08 .....	4 work-hours × \$85 per hour = \$340 .....	\$0	\$340	\$74,460
New actions .....	4 work-hours × \$85 per hour = \$340 .....	0	340	74,460

## ESTIMATED COSTS OF ON-CONDITION ACTIONS

Labor cost	Parts cost	Cost per product
2 work-hour × \$85 per hour = \$170 .....	\$500	\$670

According to the manufacturer, some or all of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected individuals. The FAA does not control warranty coverage for affected individuals. As a result, the FAA has included all known costs in the cost estimate.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative,

on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by:
  - a. Removing Airworthiness Directive (AD) 2018–07–08, Amendment 39–19239 (83 FR 15495, April 11, 2018); and
  - b. Adding the following new AD:

**2020–15–16 Leonardo S.p.A. (type certificate previously held by Agusta S.p.A.):** Amendment 39–21179; Docket No. FAA–2020–0204; Product Identifier 2018–SW–082–AD.

**(a) Effective Date**

This AD is effective August 27, 2020.

**(b) Affected ADs**

This AD replaces AD 2018–07–08, Amendment 39–19239 (83 FR 15495, April 11, 2018).

**(c) Applicability**

This AD applies to Leonardo S.p.A. (type certificate previously held by Agusta S.p.A.) Model A109E, A109K2, A109S, AW109SP, A119, and AW119 MKII helicopters, certificated in any category, with a tail rotor blade retention bolt (bolt) having part number (P/N) 709–0160–57–101 installed.

**(d) Subject**

Joint Aircraft Service Component (JASC) Code 6500, Tail Rotor Drive System.

**(e) Reason**

This AD was prompted by the discovery of a cracked bolt, and a determination that repetitive inspections of the bolt are needed to address the unsafe condition. The FAA is issuing this AD to address cracked bolts, which could result in failure of the tail rotor and loss of control of the helicopter.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

- (1) Before further flight:
  - (i) For Model A109E and A109K2 helicopters, remove from service any bolt having P/N 709–0160–57–101 that has 800 or more hours time-in-service (TIS). If the hours TIS is unknown, remove the bolt from service. Thereafter, remove from service any bolt having P/N 709–0160–57–101 before accumulating 800 hours TIS.
  - (ii) For Model A109S, AW109SP, A119, and AW119 MKII helicopters, remove from service any bolt having P/N 709–0160–57–101 that has 3,200 or more landings. If the number of landings is unknown, remove the bolt from service. Thereafter, remove from service any bolt having P/N 709–0160–57–101 before accumulating 3,200 landings. For purposes of this AD, a landing is counted anytime a helicopter lifts off into the air and then lands again regardless of the duration of the landing and regardless of whether the engine is shutdown.
  - (iii) Remove from service any bolt having P/N 709–0160–57–101 that has 800 or more hours TIS, or 3,200 or more landings, that has been interchanged between different model helicopters listed in paragraphs (g)(1)(i) and (ii) of this AD. If the hours TIS or number of landings is unknown, remove the bolt from service. Thereafter, remove from service any bolt having P/N 709–0160–57–101 that has been interchanged between different model helicopters listed in paragraphs (g)(1)(i) and (ii) of this AD before accumulating 800 hours TIS or 3,200 landings, whichever occurs first.
- (2) Within 25 hours TIS after the effective date of this AD, and thereafter at intervals not to exceed 200 hours TIS, remove each bolt having P/N 709–0160–57–101. Prior to cleaning, using a 10X or higher power magnifying glass, inspect each bolt having P/N 709–0160–57–101 for any crack in the area depicted in Figure 1 of Leonardo Helicopters Mandatory Bollettino Tecnico No. 109EP–

149, 109K-72, 109S-072, 109SP-105, or 119-080, all dated August 19, 2016, as applicable to your model helicopter.

(i) If there is any crack, replace the bolt with an airworthy bolt before further flight.

(ii) If there are no cracks, before further flight, clean and degrease the inspection area of the bolt with solvent, and using a 10X or higher power magnifying glass, inspect each bolt having P/N 709-0160-57-101 for any crack in the area depicted in Figure 1 of Leonardo Helicopters Mandatory Bollettino Tecnico No. 109EP-149, 109K-72, 109S-072, 109SP-105, or 119-080, all dated August 19, 2016, as applicable to your model helicopter. If there is any crack, replace the bolt with an airworthy bolt before further flight.

(3) As of the effective date of this AD, installation of a bolt having P/N 709-0160-57-101 is allowed, provided that the bolt has passed an inspection as required by paragraph (g)(2) of this AD.

#### (h) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Rotorcraft Standards Branch, FAA, may approve AMOCs for this AD. Send your proposal to: Matt Fuller, AD Program Manager, Continued Operational Safety Branch, Airworthiness Products Section, General Aviation and Rotorcraft Unit, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone (817) 222-5151; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, notify your principal inspector or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

#### (i) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information European Aviation Safety Agency (now European Union Aviation Safety Agency) (EASA) Emergency AD 2016-0173-E, dated August 24, 2016. This EASA AD may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0204.

(2) For more information about this AD, contact Matt Fuller, AD Program Manager, Continued Operational Safety Branch, Airworthiness Products Section, General Aviation and Rotorcraft Unit, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone (817) 222-5151; email [matthew.fuller@faa.gov](mailto:matthew.fuller@faa.gov).

#### (j) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following service information was approved for IBR on April 26, 2018 (83 FR 15495, April 11, 2018).

(i) Leonardo Helicopters Mandatory Bollettino Tecnico No. 109EP-149, dated August 19, 2016.

(ii) Leonardo Helicopters Mandatory Bollettino Tecnico No. 109K-72, dated August 19, 2016.

(iii) Leonardo Helicopters Mandatory Bollettino Tecnico No. 109S-072, dated August 19, 2016.

(iv) Leonardo Helicopters Mandatory Bollettino Tecnico No. 109SP-105, dated August 19, 2016.

(v) Leonardo Helicopters Mandatory Bollettino Tecnico No. 119-080, dated August 19, 2016.

(4) For service information identified in this AD, contact Airbus Helicopters, 2701 N. Forum Drive, Grand Prairie, TX 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <https://www.airbus.com/helicopters/services/technical-support.html>.

(5) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy, Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222-5110.

(6) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on July 16, 2020.

**Lance T. Gant,**

*Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2020-15811 Filed 7-22-20; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA-2020-0334; Product Identifier 2020-NM-014-AD; Amendment 39-21165; AD 2020-15-02]**

**RIN 2120-AA64**

#### **Airworthiness Directives; Gulfstream Aerospace LP Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Gulfstream Aerospace LP Model Gulfstream G280 airplanes. This AD was prompted by a report of inadequate clearance between the fuel probes and forward fuel tank structure. This AD requires measuring the clearance between certain fuel probes and the forward fuel tank structure, and reinstalling the probes if necessary, as specified in a Civil Aviation Authority

of Israel (CAAI) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective August 27, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of August 27, 2020.

**ADDRESSES:** For the material incorporated by reference (IBR) in this AD, contact the CAAI, P.O. Box 1101, Golan Street, Airport City, 70100, Israel; telephone 972-3-9774665; fax 972-3-9774592; email [aip@mot.gov.il](mailto:aip@mot.gov.il). You may find this IBR material on the CAA website at [www.caa.gov.il](http://www.caa.gov.il). You may view this IBR material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0334.

#### **Examining the AD Docket**

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-0334; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206-231-3226; email [tom.rodriguez@faa.gov](mailto:tom.rodriguez@faa.gov).

#### **SUPPLEMENTARY INFORMATION:**

##### **Discussion**

The CAAI, which is the aviation authority for Israel, has issued Israeli AD ISR-I-53-19-10-5, dated October 10, 2019 ("Israeli AD ISR-I-53-19-10-5") (also referred to as the Mandatory Continuing Airworthiness Information, or "the MCAI"), to correct an unsafe condition for certain Gulfstream Aerospace LP Model Gulfstream G280 airplanes.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR